

Experimental Analysis and Modeling of Z-pinned Joints under Pull-out, Shear and Flexure Loadings

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Overview

- Introduction
- Pinned T and L joints
- Experimental
- Modeling
- Conclusion

Z-pinning technology

Z-pinning in literature:

- Main research goal: Increase delamination strength of laminates
- Most widespread manufacturing technique: Z-fiber[®]



Laminate reinforced with z- pinning (z-fiber®)



Z-pinned « Lap-joint »

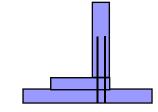
Present work

Low-cost laminates joint by z-pinning (ALCAS program)



- Twisted carbon pins
- $\bullet \emptyset = 0.75mm$
- L=40mm





 Research goal: provide physical understanding and mechanical modeling of pinned joints

Overview

Introduction

Pinned T and L joints

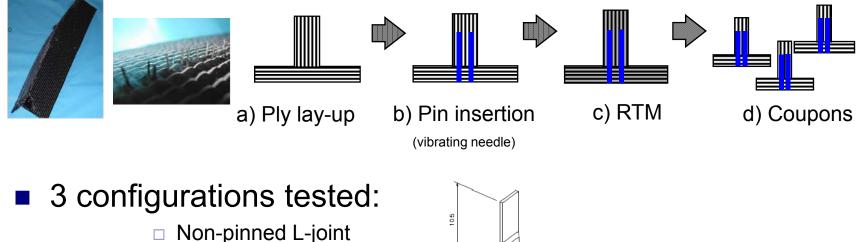
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Pinned T and L joints

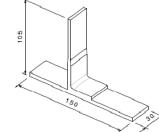
- Joints investigated:
 - Skin/stiffener joint: typical in aircraft structures

Manufacturing:

(EADS IW + Dassault Aviation) [1]



- Pinned L-joint
- □ Pinned T-joint



[1] Lefebure P. Experimental Assembly of dry Carbon Preforms By a Z pinning technology. SAMPE Europe Technical Conference. SETEC 01/06

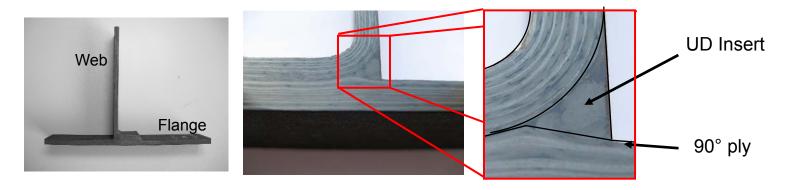
Non-pinned L-joint:

□ Web:

- Thickness: 3.8mm
- Material: NCF PRIFORM HTS 450 [[90°/+45°/-45°]_n]_s

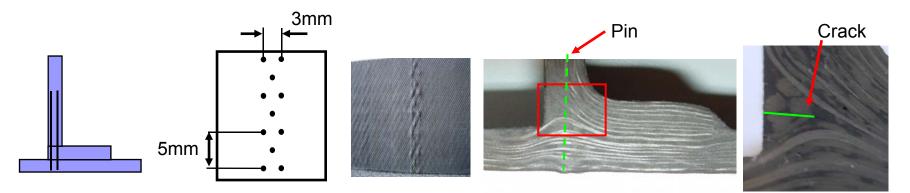
□ Flange:

- Thickness: 5.3mm
- Material: Uniwave PRIFORM IMS 200, quasi-isotropic sequence



Pinned L-joint:

- □ Web / Flange jointed by carbon pins (18 pins/coupon)
- Same materials as non-pinned L-joint
- Pins produce increase in web thickness and waviness on flange plies
- Cracks have been detected over the UD insert region



Pinned T-joint:

No web foot: Web / Flange ONLY joint by carbon pins (18 pins/coupon)



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Experimental analysis

- 3 Static loadings (I):
 - Pull-out



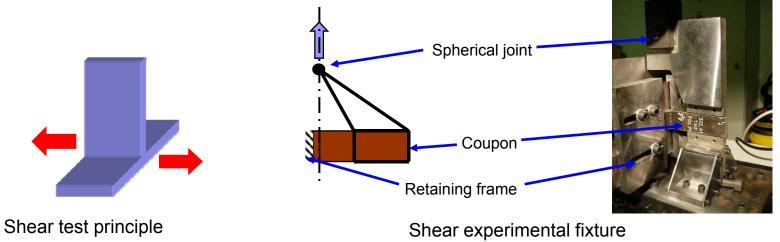
Pull-out test principle



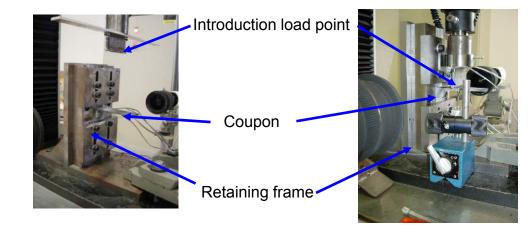


Pull-out experimental fixture



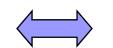


- 3 Static loadings (II):
 - Flexure

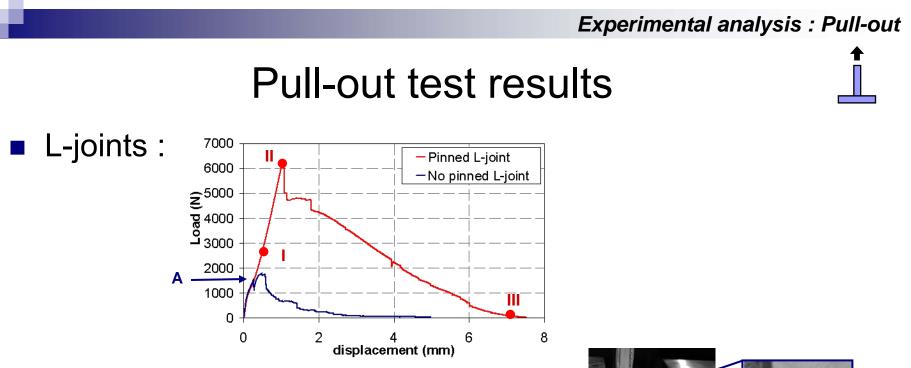


Flexure test principle

- Parameters to analyze:
 - □ Loads:
 - Limit load (1st failure load)
 - Ultimate load
 - Pin behavior



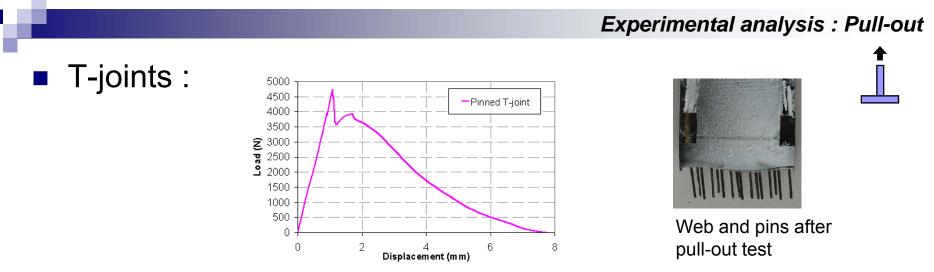
Application on aircraft structures



- □ Non-pinned L-joint:
 - Failure by interface delamination (A)
- A

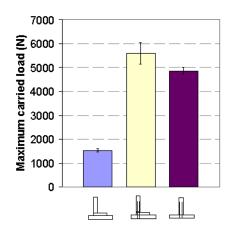
- □ Pinned L-joint
 - Crack opening (I)
 - Pins debond from flange (II)
 - Pins slip from flange (III)





□ Failure: Pins debond from flange + residual friction

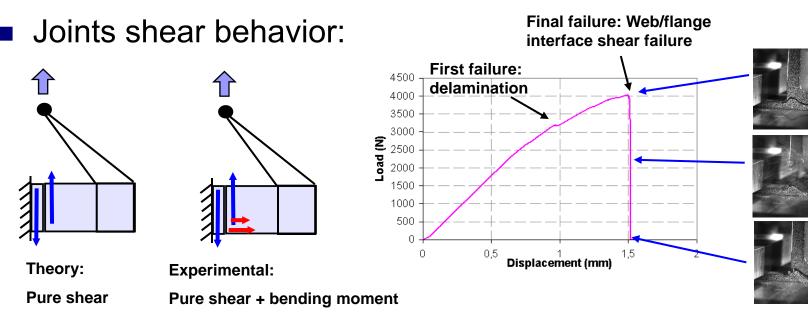
Maximum carried load comparison



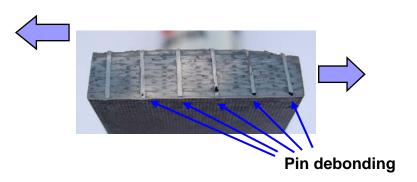
- Pinned L-joints and T-joints strength +260% and +215% higher than Non-pinned L-joint
- Pinned joints: same failure mode in L and T- joints (pins debond from flange)

Experimental analysis : Shear

Shear test results



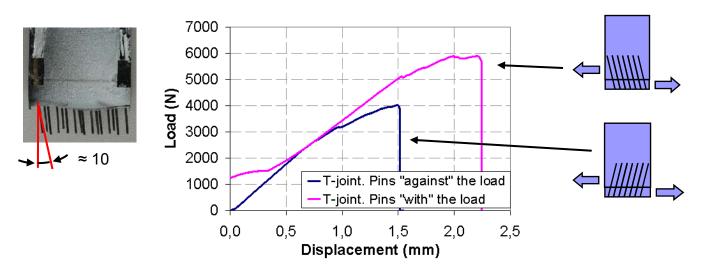
□ Pins behavior: pin debonding from flange + shear failure



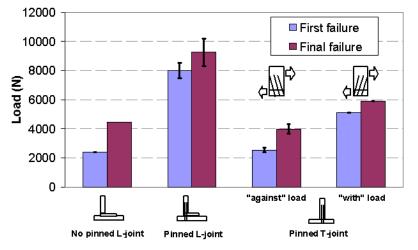
Experimental analysis : Shear



Pins orientation dependence of pinned T-joint :

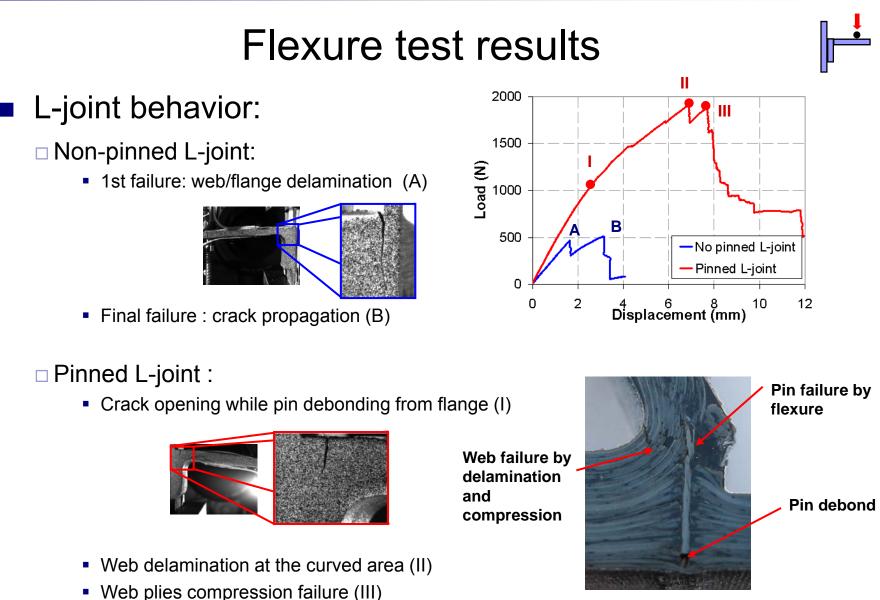


Joint strength comparison

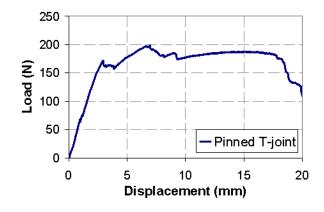


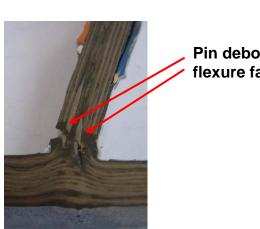
- L-joints: Pins increase both first failure and final failure load
- T-joints :Strong strength dependence on pins orientation. +48% if oriented "with" load
- □ Specimen sudden failure by shear at the web/flange interface → Test bench OK

Experimental analysis : Flexure



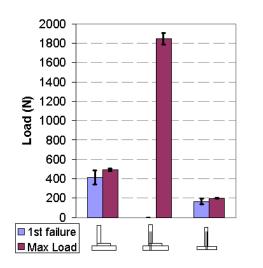
Pinned L-joint post-mortem analysis





Pin debonded + flexure failure

Strength comparison



- Pins increase strongly maximum carried load level on L-joints but difficult to establish 1st failure level
- T-joint strength much lower than L-joint due to their architecture (low bending stiffness at web base)

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Modeling introduction

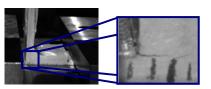
Phenomena to be modeled:

□ Origin of cracks at the web/flange interface in pinned joints



Mechanical behavior

Non-pinned joints: 1st failure criteria under Pull-Out and Flexure

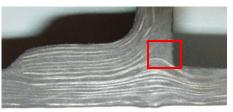


- Pinned joints:
 - Pin modeling
 - Test simulation (Pull-Out / Shear/ Flexure)
 - For predicting strength
 - To estimate limit load

Thermal loading on pinned joints

Specimens observations:

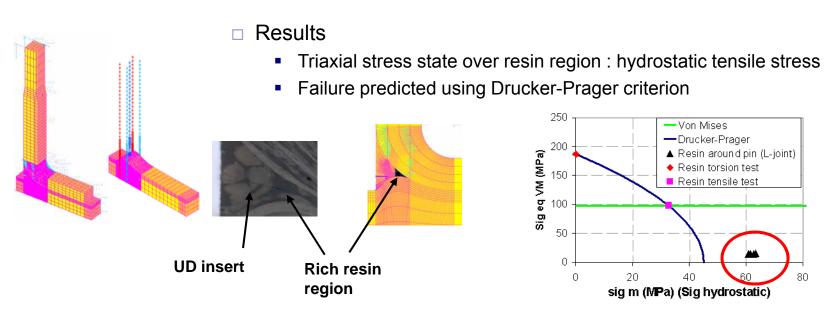
Cracks at the rich resin regions around pins at the web/flange interface





\Box Model:

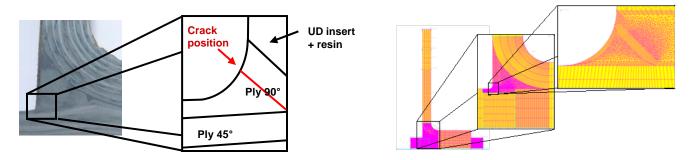
Thermal loading after curing cycle: (ΔT=-155°C)



Non-pinned joints modeling

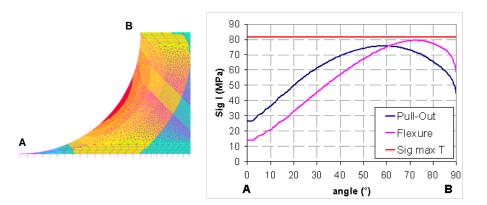
2D FE model for pull-out and flexure simulation:

□ Model: High detail resolution of the coupons geometry



□ Stress state at the 1st failure load level :

• High stress concentration (pure tensile) : Criterion on $\sigma_{t max}$ of ply 90°



 Model 1st failure load level prediction: +7% pull-out

+1% flexure



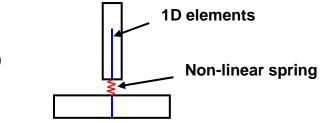
- Criterion OK
- Key factor: high detail resolution

Modeling : pinned joints

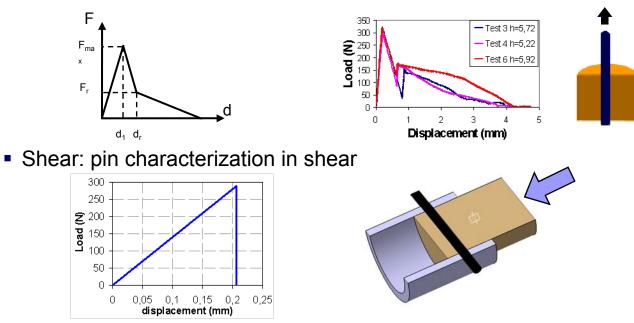
Pinned joints under mechanical loadings



- 1D elements (beams) into laminates
- Non-linear springs at the interface (pull-out and shear)



- Pin behavior
 - Tensile (pull-out) : based on single pin pull-out test bonded in net resin [2]



[2] Toral Vazquez J, Castanié B, Barrau JJ, Didierjean S, *Experimental analysis and modeling of Z-pinned joints under pull-out loading* Proceedings of ECCM13, 2008

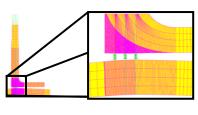
L-joint under pull-out and flexure loading

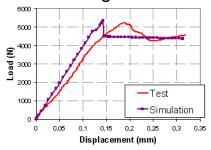
□ Joint model:

- 2D model
- Pull-out :only pin behavior at the web/flange interface (no resin layer)
- Flexure: crack through 10mm at the interface (cracks already present on the joints)

□ Pull out:

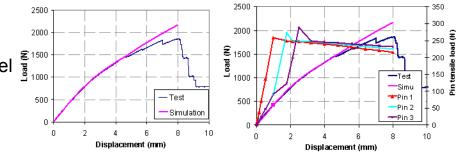
Predicted strength 5% lower than experimental strength

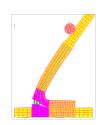




□ Flexure:

- Good agreement in load/displacement curve
- Final failure not modeled: not pins failure but web failure
- Model can estimate 1st failure load level

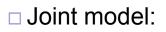




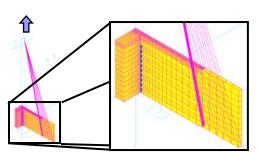
Modeling: Shear

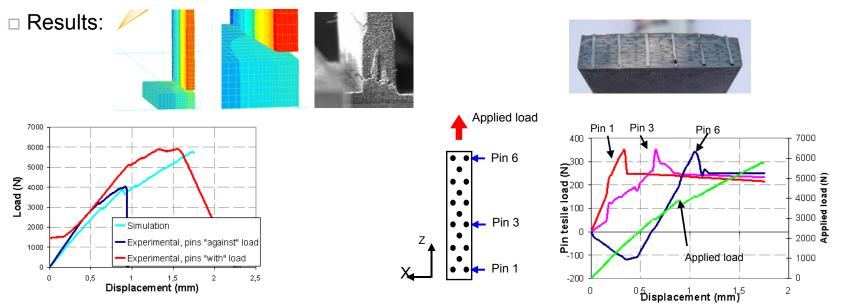
T-joint under shear loading





- 3D model
- No resin layer at the interface
- Contact between flange and web
- Pins as no linear springs





- Joint failure by pins debonding + shear failure
- Joint strength predicted if pins under tensile load
- Still misunderstanding of joints with pins "againts" load : IN PROGRESS

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Experimental analysis:

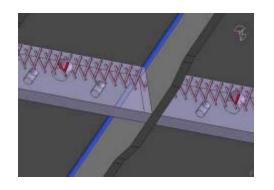
- Pins increase pull-out and flexure strength (until +260%) in L-joints
- Pinned T-joints +200% stronger than Non-pinned L-joints under pullout
- Pinned T-joint as resistant as Non-pinned L-joins in shear
- Pins loaded mostly in tension even while joint under shear loading
- Modeling:
 - 1st failure criterion of Non-pinned joints
 - Thermal analysis showing cracks origin into rich resin regions
 - Mechanical modeling of pinned joints:
 - □ Good agreement under pull-out, flexure and shear loadings:
 - Helpful to provide joint behavior understanding
 - Able to predict joint strength in most of cases
 - □ Still progress to do in mixed-loadings

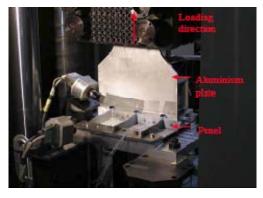
Future works

• Testing:

- Fatigue loadings
- Ageing

Application in demonstrator







THANK YOU !!!



UE FP6 ALCAS project

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